

# UAV-Based Weather Sounding Platform For Precision Weather Modification Planning and Operation

Intelligent Structures and Automatic Control Laboratory and Center for Unmanned Aerial Vehicles of Kasetsart University (ISAAC LAB and KU UAV)

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## Aerospace Engineering Technolgy@ISAAC LAB – KU UAV

Advanced Technology for Our Country Development

Design for Practical Civilian Applications

Commercial-Grade Manufacturing and Integration

Professional Standard Operation

### ISAAC LAB – KU UAV **UAS Development Roadmap**

- Key advanced research areas
  - Aircraft design
  - Smart structures & materials
  - Automatic adaptive control

Composite manufacturing

**Autonomous** small UAV systems with practical applications

(Artificial) Intelligent autonomous interactive swam UAV systems with practical interactive swam applications

Autonomous

**UAV** systems with practical

applications

Autonomous mini UAV systems with civilian applications

#### Project objectives

- Unmanned Aerial Vehicle(UAV) technology is becoming more significance in many aspects of science. Weather modification operation of the department of Royal Rain Making could also benefit from the such technology. This is the pilot project to demonstrate the possibility of using unmanned aircraft system (UAS) in weather sounding and in-situ atmospheric measurement.
- In this project, the systems is tasked to fly up to 20000 ft.
   Equipped high precision weather sounding equipment, the aircraft will provide the scientists with real-time data feed operation planning.

### **UAS for Aerial Sounding**

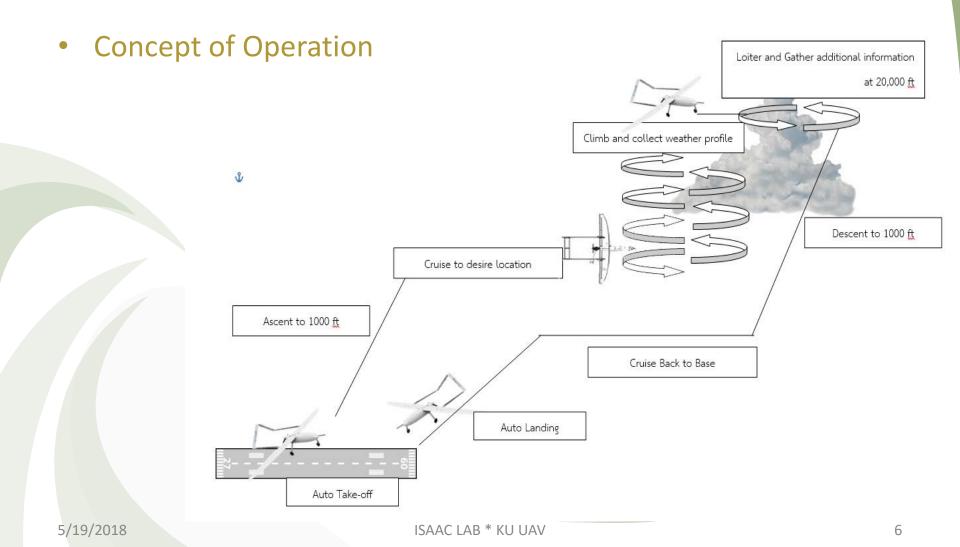
Development Framework

High Altitude
Aerial Sounding
UAS Design and
Manufacturing

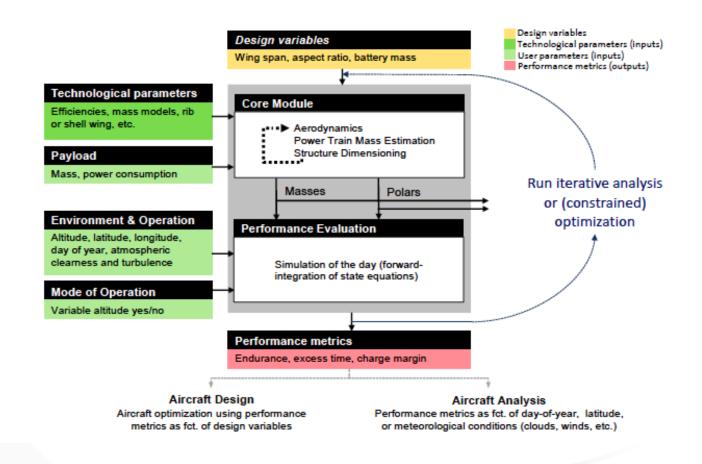
Component level tuning, testing and system level tuning, testing

Weather sounding experiment and verification with weather balloon

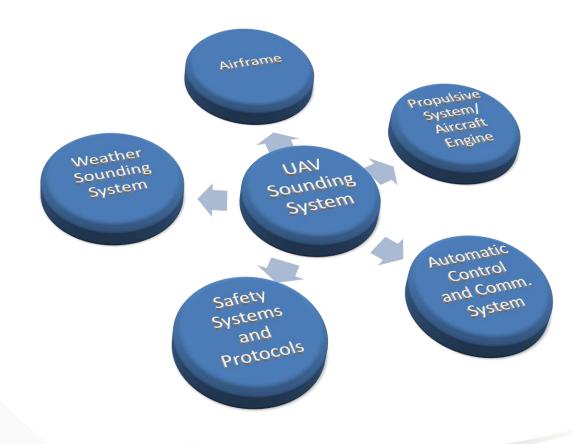
#### Aerial Sounding UAS



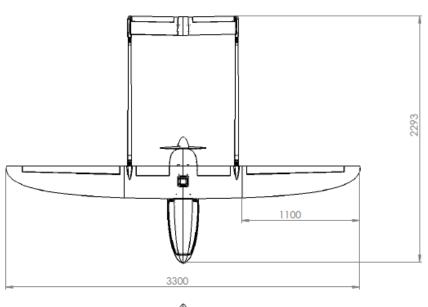
#### **UAS Design Methodology**



#### **UAS Components**

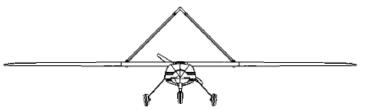


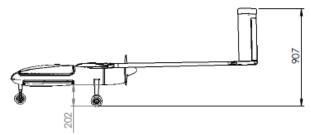
## Airframe design (specifications and performance)



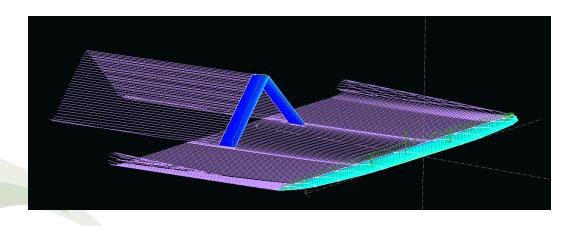
MTOW 21.5 kg  Empty weight 10 kg  Wing span 3.3 m  Length 2.27 m  Wing area 0.79 m²  Power plant 2.5 hp	Spec.'s	value
Wing span 3.3 m  Length 2.27 m  Wing area 0.79 m <sup>2</sup> Power plant 2.5 hp	MTOW	21.5 kg
Length 2.27 m  Wing area 0.79 m <sup>2</sup> Power plant 2.5 hp	Empty weight	10 kg
Wing area 0.79 m <sup>2</sup> Power plant 2.5 hp	Wing span	3.3 m
Power plant 2.5 hp	Length	2.27 m
·	Wing area	0.79 m <sup>2</sup>
	Power plant	2.5 hp
Max. payload 10 kg	Max. payload	10 kg

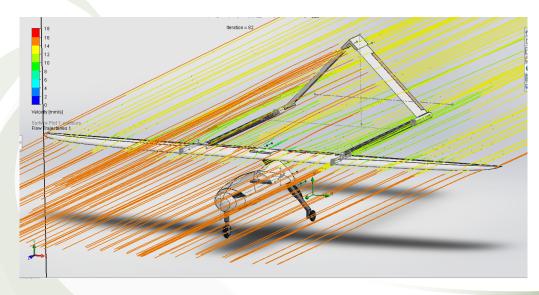
Performance	value
Endurance	8 hours
Cruise speed	22 m/s
Stall speed	13 m/s
Max. speed	36 m/s
Take-off run	30 m
CL max.	1.3
CL max. (45° flap)	1.7





#### Airfeame design and basic analysis





### Airframe manufacturing

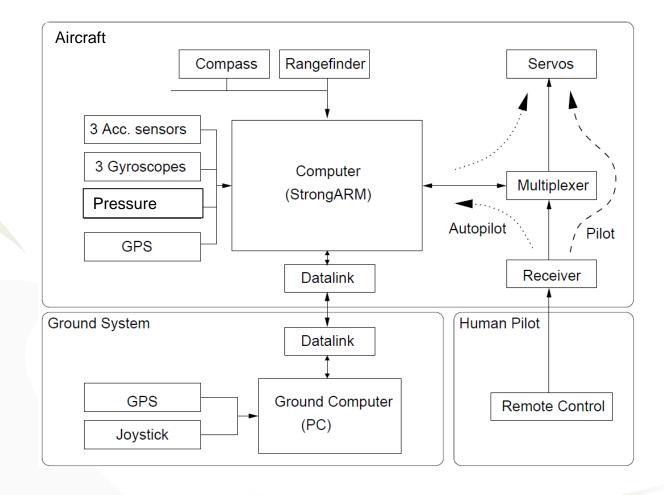


#### Propulsion system/aircraft engine

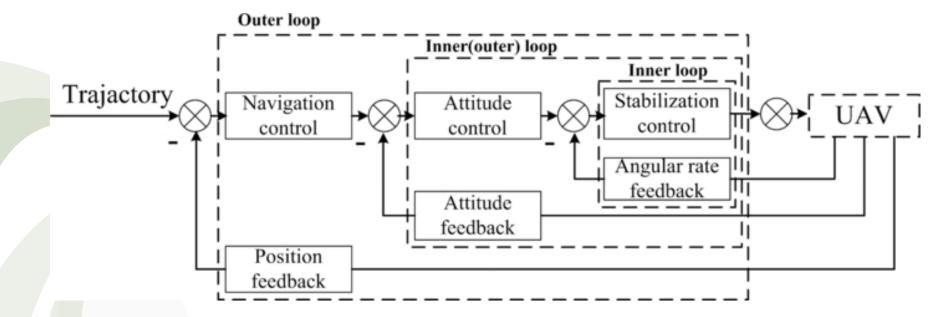
Specifications	value
Engine Type	EFI two-stroke, single cylinder, air cooled
Power	2.5 kW (3.4 hp)
Fuel consumption	400 g/kWh in cruise
Fuel type	Automotive 95+ octane
Displacement	28 cc (1.74 cu in)
Speed range	1600-8500 rpm
Recommended 2-blade propellers	16x10, 18x8, 18x10, 19x11, 20x8
Engine weight (including generator, cooling system, servo, air filter )	1.5 kg
Muffler	Combined reactive- dissipative
Sound level, measured from 30 m distance	59 dB



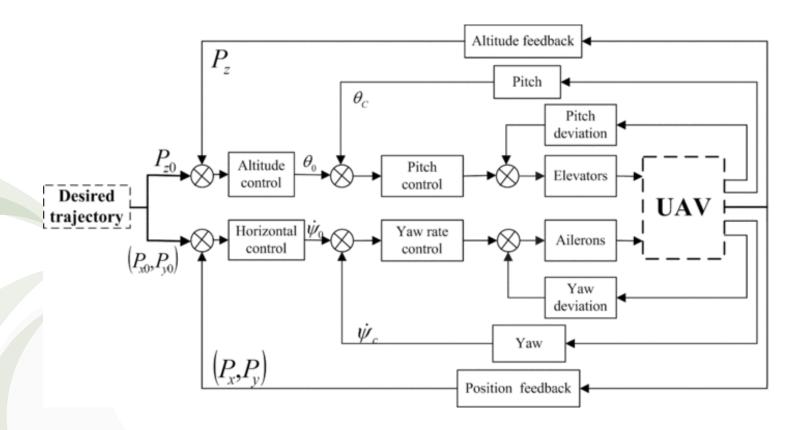
## Automatic control concepts and communication systems



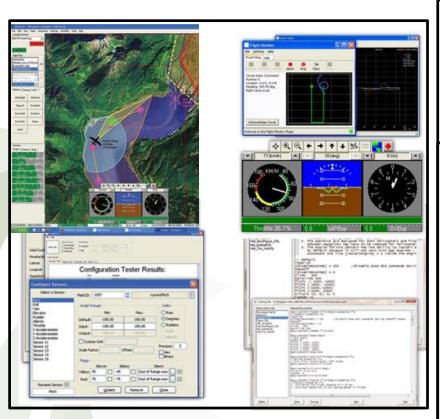
#### Aircraft control algorithm (low level)



#### Navigation algorithm



### Ground control software, hardware and communication systems







Flight Control Computer

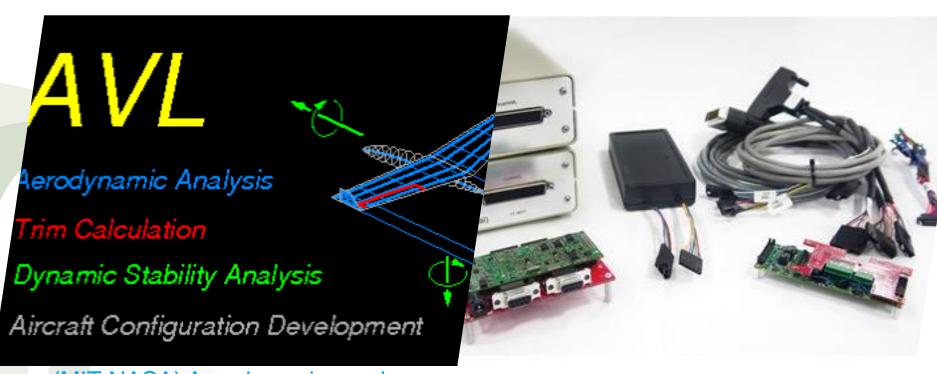
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**AGL** 



#### Hardware in the loop (HIL) testing

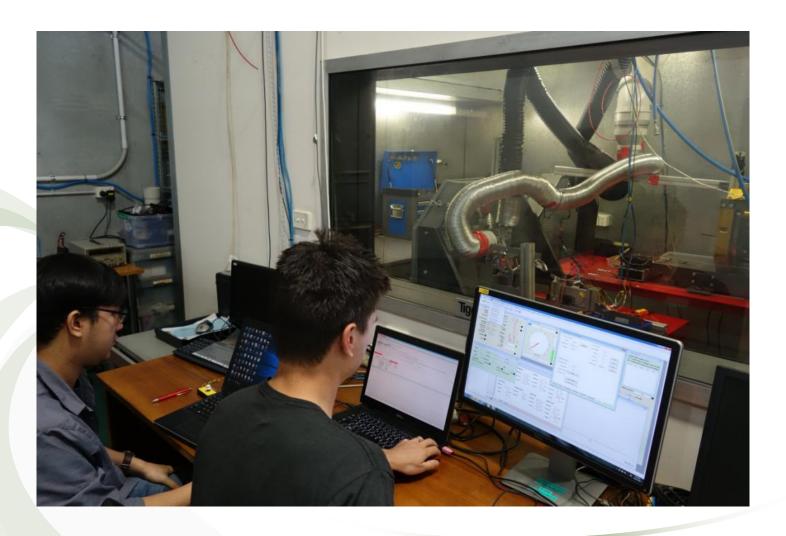


(MIT-NASA) Aerodynamics and Stability Analysis Software

## Automatic control system testing on a scaled-model aircraft



#### Propulsion system testing



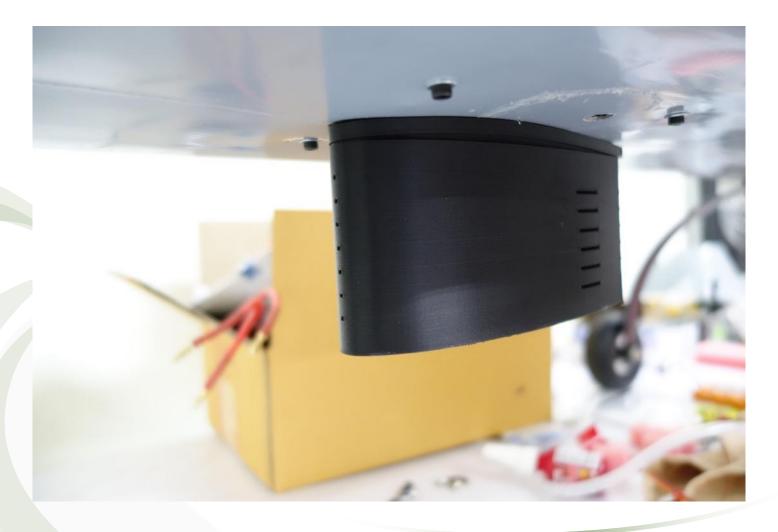
## Compatibility tuning of automatic control and propulsion systems



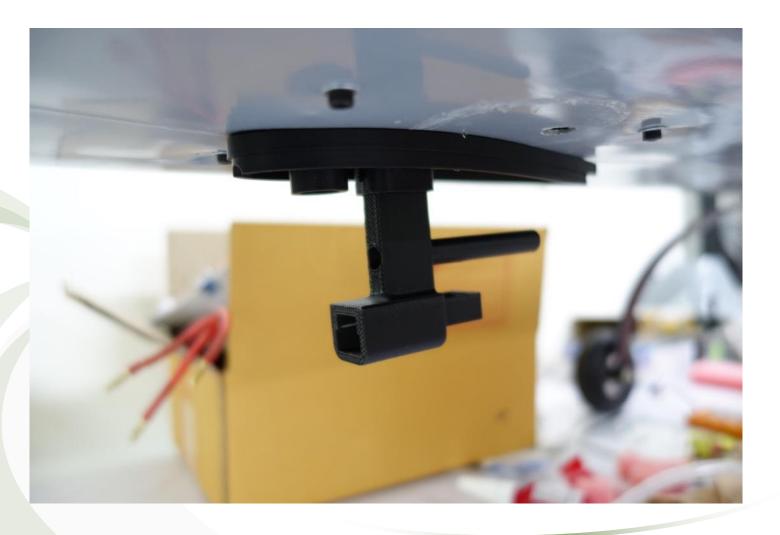
#### Flight testing at Nakorn Sawan Airport



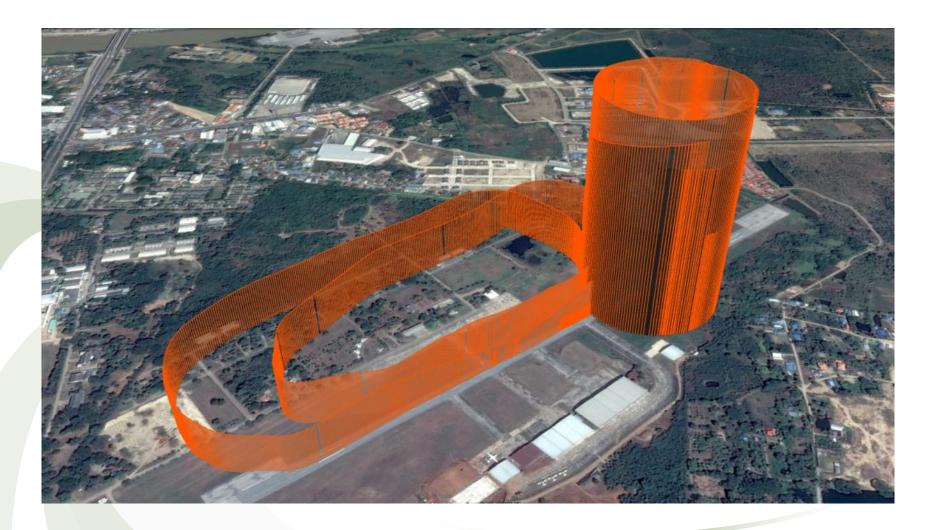
#### Weather sounding system



#### Weather sounding system



### Example flight path



#### Current State of the Development

- The systems has finished the testing to the service altitude of 9000 ft.
- To go further, nevertheless, the development need to acquire the clearance from the related authority and regulator (CAAT, etc.)
- This project will be marked as a historic point where such a civilian unmanned aircraft can be performed in the commercial-airspace.

#### Future Development Possibilities

- The know-how and knowledge from this pilot project will lead to the new paradigm of weather modification operation.
- The systems can be enhanced to fly higher and retrofit to perform different task such as tropical-weather research or chemical-flare deploy as well as information gathering for further break-through in tropical region weather modification research.
- The development team aims to equip the aircraft with Silverlodine flare for high altitude rain making application.



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